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An Improved Series Equivalent Circuit of a Single-Sided Linear Induction Motor

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Abstract- The derivation of the equivalent circuit for a single-sided linear induction motor (SLIM) is not straightforward. Especially if it includes longitudinal end effects from the cut-open primary magnetic path, transverse edge effects from the differing widths between the primary lamination and secondary sheet, and half filled primary slots. This paper proposes an improved series equivalent circuit for this machine. The longitudinal end effects are estimated using three different impedances representing the normal, forwards and backwards flux density waves in the air-gap. The transverse edge effects accounted for with a correction coefficient K_t and an air-gap flux density correction coefficient K_b . Using the series circuit, the performance of the SLIM was assessed in a similar manner to a rotating induction machine. A 4 kW SLIM prototype was tested which validated the simulation technique.

I. INTRODUCTION

Single-sided linear induction motors (SLIMs) are derived from rotating induction motors (RIMs). They can be utilized in a number of industrial systems; they have obvious application in rapid light transportation and even in very low speed low-head hydro (water-wheel) generation (where the copper or aluminum secondary plate is on the large-diameter wheel). In transportation systems, they have the advantage (over an RIM) of the propulsion thrust being achieved without reliance on friction between wheel and rail (allowing high inclines) as well as several others [1, 2]. While most current literature for new transport or renewable energy applications consider various forms of permanent-magnet machine for the electromechanical energy conversion unit, the linear machine may have niche applications in these areas.

In a rotating machine, an accurate equivalent circuit model can be obtained with reasonable ease because of pole symmetry. The model of the linear equivalent is more complicated because some parameters in the circuit are asymmetrical [3], i.e., it has transverse and longitudinal end effects. The transverse edge effects result in changes in the transverse air-gap flux density. The longitudinal end effects are derived from the primary discontinuous magnetic circuit where the electrical conditions can vary greatly. When a double layer winding is used some of the end slots will only contain one coil side so that they will be only half filled.

There have been many studies conducted on the SLIM; many make approximations. For example, exit end-wave analysis is not considered in [3], half-filled slots are not taken into account in [4], and transverse end effects are ignored in [5]. Such assumptions generate errors in the analysis of the SLIM performance. This paper uses the air-gap flux density

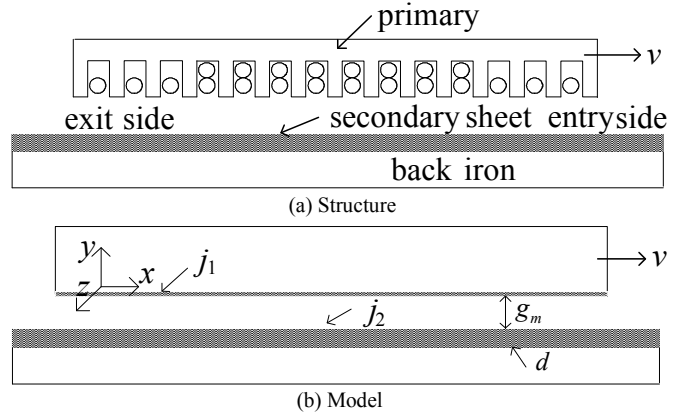


Fig. 1. The structure and one-dimensional analytic model of SLIM. (a).Structure.(b).Model.

equation develop an improved series equivalent circuit which considers the influences of the entry and exit end-wave flux waves in the air-gap, as well as transverse effects, half filled slots and air-gap leakage flux.

II. THE PHYSICAL MODEL

To simplify the analysis, some assumptions are used [6-12]: (a) the stator iron has infinite permeability; (b) skin effect is neglected in the secondary; (c) winding space harmonics are negligible; and (d) the primary and secondary currents flow in infinitesimally thin sheets.

The structure and one-dimensional representation of the SLIM are shown in Fig. 1. The model described in [11] derives several relevant coefficients for the secondary sheet current and air-gap flux density. The analysis from [12] can be used to analyze the system. From Ampere's Law (with reference to the geometry in Fig. 1):

$$\frac{g}{\mu_0} \frac{\partial b_y}{\partial x} = j_1 + j_2 \quad (1)$$

We can implement a one-dimensional Curl function for the secondary plate - the air-gap electric field has to be referred to the secondary:

$$\frac{\partial e_z'}{\partial x} = \frac{\partial b_y}{\partial t} + \frac{\partial b_y}{\partial x} \frac{\partial x}{\partial t} = \frac{\partial b_y}{\partial t} + v \frac{\partial b_y}{\partial x} \quad (2)$$

where the second term on the right hand side is the speed term. What we now want to try and do is refer the secondary current sheet to the primary so that we can obtain an expression for the air-gap flux density in terms of the (measureable) primary current. The secondary consists of a

copper conducting sheet and we will assume that the current is resistance-limited so that for a plate of thickness d and surface conductivity σ :

$$e_z' = \frac{j_2}{\sigma} \quad (3)$$

where $\sigma = d/\rho$ (in ohm) and ρ is the volume resistivity (in ohm-m). We can combine these equations so that

$$g \frac{\partial^2 b_y}{\partial x^2} - \sigma \mu_0 v \frac{\partial b_y}{\partial x} - \sigma \mu_0 \frac{\partial b_y}{\partial t} = \mu_0 \frac{\partial j_1}{\partial x} \quad (4)$$

If we assume that

$$j_1 = j_1(x, t) = \text{Re} \left\{ \bar{J}_1 e^{j(\omega t - \pi x / \tau)} \right\} \quad (5)$$

then we can obtain an expression for the air-gap flux density:

$$g \frac{\partial^2 b_y}{\partial x^2} - \sigma \mu_0 v \frac{\partial b_y}{\partial x} - j \sigma \mu_0 \omega b_y = \text{Re} \left\{ -j \mu_0 \frac{\pi}{\tau} \bar{J}_1 e^{j(\omega t - \pi x / \tau)} \right\} \quad (6)$$

The solution of b_y is

$$b_y = b_y(x, t) = \text{Re} \left\{ \left[\bar{B}_0 e^{-j \frac{\pi x}{\tau}} + \bar{B}_1 e^{-\frac{x}{\alpha_1}} e^{-j \frac{\pi x}{\tau_e}} + \bar{B}_2 e^{\alpha_2} e^{j \frac{\pi x}{\tau_e}} \right] e^{j \omega t} \right\} \quad (7)$$

We can derive the air-gap flux density coefficients so that

$$\bar{B}_0 = \frac{j \tau \mu_0 \bar{J}_1}{g \pi (1 + j s G)} \quad (8)$$

$$\text{where } G = \frac{\sigma \mu_0 \omega \tau^2}{g \pi^2}$$

and the slip $s = (v_s - v)/v_s$; we need to define the synchronous velocity $v_s = \omega \tau / \pi$ with pole pitch τ . The second term is

$$\bar{B}_1 = j \frac{\tau}{\pi} \left(\frac{1}{\alpha_1} + j \frac{\pi}{\tau_e} \right) \bar{B}_0 \quad (9)$$

$$\text{where } \alpha_1 = \frac{2g}{Xg - \sigma \mu_0 v} \text{ and } \tau_e = \frac{2\pi}{Y}.$$

$$\text{We can define } X + jY = \sqrt{\left(\frac{\sigma \mu_0 v}{g} \right)^2 + 4j \frac{\sigma \mu_0 \omega}{g}}.$$

In a similar fashion

$$\bar{B}_2 = -j \frac{\tau}{\pi} e^{-\left(\frac{1}{\alpha_2} + j \frac{\pi}{\tau_e} \right) L_p} \left(\frac{1}{\alpha_2} + j \frac{\pi}{\tau_e} \right) \bar{B}_0 \quad (10)$$

$$\text{where } \alpha_2 = \frac{2g}{Xg + \sigma \mu_0 v}.$$

We now have a complete definition of the air-gap flux-density wave in terms of the stator surface current density. Most parameter definitions from (1) to (10) are given in the appendix. In (2), the flux density b_y includes three parts: b_0 , b_1 and b_2 . b_0 is the normal traveling wave which moves forward in a similar manner to the fundamental flux density wave in a rotating induction machine (RIM). b_1 and b_2 are determined from the boundary conditions, which are the entrance and exit end-effect waves [9-11]. b_1 is a gradually

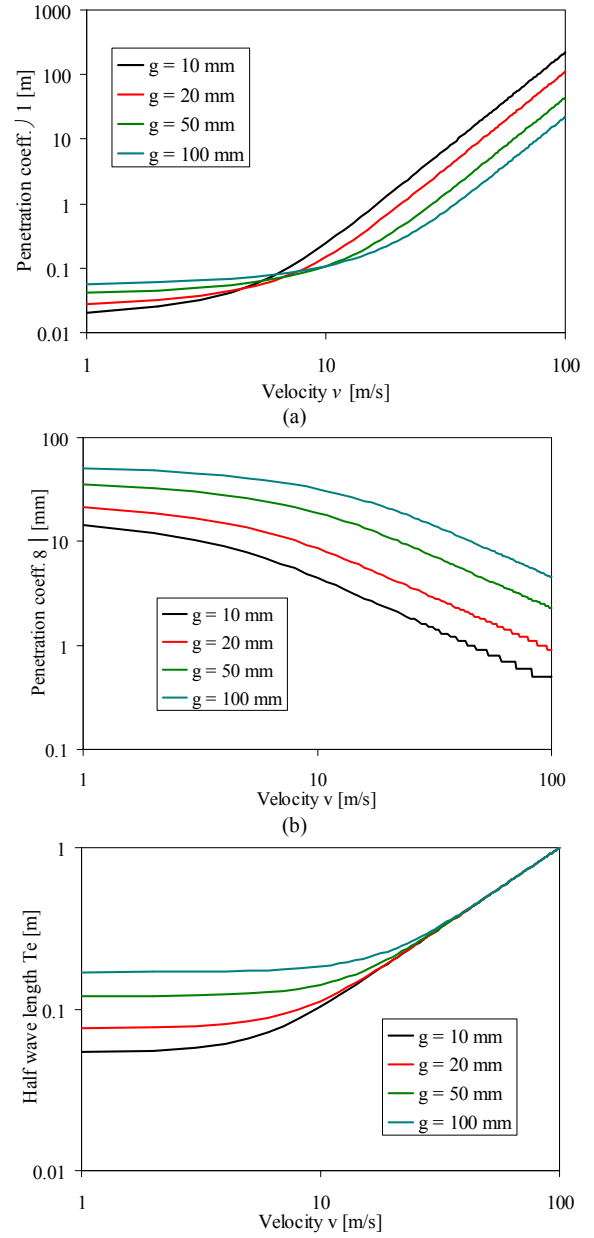


Fig. 2. Penetration coefficients and half-wave length versus motor velocity. (a) Entry penetration coefficient α_1 . (b) Exit penetration coefficient α_2 . (c) Half-wave length τ_e .

attenuating wave traveling along the x -axis. Its attenuation constant is $1/\alpha_1$ and end-effect half-wave length is τ_e . b_2 travels along the x -axis in the negative direction with an attenuation constant of $1/\alpha_2$ [10].

The curves for α_1 , α_2 and τ_e are shown in Fig. 2. It is assumed that the SLIM operates from zero to 100 m/s, and the secondary copper sheet equivalent surface resistivity ($=\rho/d$) is 5.63×10^{-5} ohm where the plate thickness d is 3 mm. From Fig. 2 (a), the length of the entry wave penetration coefficient α_1 increases with motor velocity v . It is close to 0.4 m when equivalent air-gap length g is 10 mm and v is 10 m/s. However, the length of the entry wave penetration coefficient α_2 in Fig. 2 (b) decreases with increasing motor velocity, and it is almost 2 mm when g is 10mm and v is 10

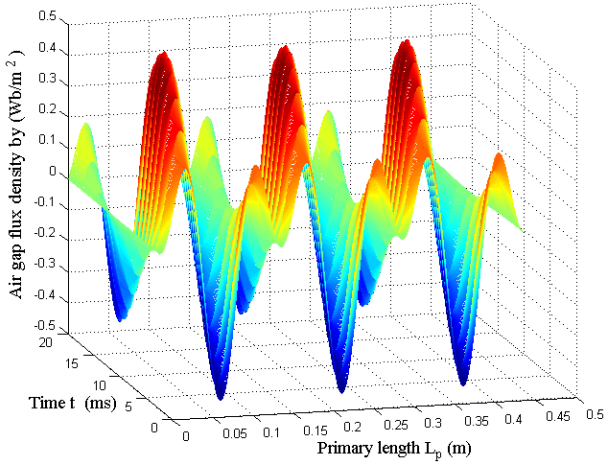


Fig. 3. Air gap flux density distribution in the primary length (6-pole wave progression illustrated).

m/s. From Fig. 2 (c), the half-wave length τ_e of the end-effect wave increases with velocity, is close to 0.1 m when g is 10 mm and v is 10 m/s.

Fig. 3 shows a SLIM air-gap flux density distribution for the machine studied here over the primary length area. The final flux wave is not sinusoidal because of the influence of the entry and exit flux waves.

The electric field intensity in the air-gap is denoted by (2). However this now needs to be referred to primary so that the speed term is ignored. Hence

$$e_z(x, t) = \int \frac{\partial b_y}{\partial t} dx = \text{Re} \left\{ \begin{array}{l} \bar{E}_0 e^{j(\omega t - \frac{\pi x}{\tau})} + \bar{E}_1 e^{-\frac{x}{\alpha_e}} j \left(\omega t - \frac{\pi x}{\tau_e} \right) \\ + \bar{E}_2 e^{\frac{x}{\alpha_e}} j \left(\omega t + \frac{\pi x}{\tau_e} \right) \end{array} \right\} \quad (11)$$

With knowledge of the air-gap flux density distribution, the electric field intensity produces three terms: E_0 , E_1 and E_2 . These individually produce an induced electromotive force in the machine windings so that

$$\bar{E}_0 = -\omega \frac{\tau}{\pi} \bar{B}_0 = -v_s \bar{B}_0 \text{ and}$$

$$\bar{E}_1 = -\frac{j\omega \bar{B}_1 \alpha_1 \tau_e}{(\tau_e + j\pi \alpha_1)}, \bar{E}_2 = \frac{j\omega \bar{B}_2 \alpha_2 \tau_e}{\tau_e + j\pi \alpha_2}.$$

III. EQUIVALENT CIRCUIT

The equivalent circuit is essential for analyzing the motor and predicting its performance. It is particularly necessary for the initial SLIM electromagnetic design procedure. Here the induced electromotive force is used to compute the fundamental-wave impedance. The complex power and electric-magnetic thrust methods are adapted to derive the entrance- and exit-wave impedances, which are utilized to calculate the SLIM thrust, power factor, efficiency and other performance parameters.

A. Fundamental-wave Impedance

The fundamental elements such as excitation current, electric field intensity and flux density are assumed to be

sinusoidal shown below. From [11], the entrance and exit-wave elements, such as $b_1(x, t)$, $b_2(x, t)$, $e_1(x, t)$, $e_2(x, t)$, have similar forms. We can use the following equations to represent the current, air-gap flux density and electric fields:

$$j(x, t) = \text{Re} \left\{ \bar{J}_1 e^{j\left(\omega t - \frac{\pi x}{\tau}\right)} \right\} \quad (12)$$

$$b_0(x, t) = \text{Re} \left\{ \bar{B}_0 e^{j\left(\omega t - \frac{\pi x}{\tau}\right)} \right\} \quad (13)$$

$$e_0(x, t) = \text{Re} \left\{ \bar{E}_0 e^{j\left(\omega t - \frac{\pi x}{\tau}\right)} \right\} \quad (14)$$

The magnetic motive force $F(x, t)$ is of a similar form:

$$F(x, t) = \text{Re} \left\{ j \frac{g\tau}{\pi} \bar{J}_1 e^{j\left(\omega t - \frac{\pi x}{\tau}\right)} \right\} \quad (15)$$

The relationship between primary current sheet J_1 and phase current I is

$$\bar{J}_1 = jJ_1 = \frac{\sqrt{2}mw_1K_{w1}}{\tau P} I \quad (16)$$

where we assume that the reference current is real. The induced phase electromotive force is

$$\bar{E}_0 = j\sqrt{2}w_1K_{w1}l_w\bar{E}_0 \quad (17)$$

The fundamental-wave impedance can be calculated:

$$\bar{Z}_m = \frac{\bar{E}_0}{I} = \frac{jX_m}{(1 + jsG)} \quad (18)$$

where X_m is

$$X_m = \frac{4mf(w_1K_{w1})^2l_w\tau\mu_0}{P\pi g} \quad (19)$$

The fundamental-wave thrust F_{e0} is denoted by

$$F_{e0} = \frac{l_w}{2} \text{Re} \int_0^{L_p} \bar{B}_0 e^{-j\frac{\pi x}{\tau}} \left\{ \bar{J}_1 e^{-j\frac{\pi x}{\tau}} \right\}^* dx = \frac{ml^2 \text{Re}\{\bar{Z}_m\}}{v_s} \quad (20)$$

where the integral length along the primary length $L_p = 2\tau P$. The mechanical power can be obtained from (20) by multiplying by the speed v_r . But $v_r = (1 - s)v_s$ so that the mechanical power becomes

$$P_{m0} = ml^2 \text{Re}\{(1 - s)\bar{Z}_m\} \quad (21)$$

We can investigate Z_m in (18). If we follow through some rearrangement of (18) then we obtain

$$\bar{Z}_m = \frac{jX_m}{(1 + jsG)} = \frac{jX_m \left(\frac{r_2/s}{r_2/s + jX_m} \right)}{\left(\frac{r_2/s}{r_2/s + jX_m} \right)} = \frac{jX_m}{\left(1 + sjX_m/r_2 \right)}$$

which can be represented by the equivalent circuit in Fig. 4(b). From this the rotor resistance (referred to the stator) is

$$r_2 = \frac{2m(w_1K_{w1})^2l_w}{\sigma\tau P}$$

This derivation illustrates that the theory is similar to the standard equivalent circuit analysis for rotating machines. Because the secondary is a simple cooper plate then the secondary inductance is assumed negligible and the current is resistance limited.

B. Entrance-wave Impedance

The electromagnetic power P_{m1} for the y -axis entrance flux wave is the power flowing into the rotor through this wave:

$$P_{m1} = \frac{l_w}{2} \operatorname{Re} \int_0^{L_p} -\bar{E}_1 e^{-\frac{x}{\alpha_1} - j\frac{\pi x}{\tau_e}} \cdot \left\{ \bar{J}_1 e^{-j\frac{\pi x}{\tau}} \right\}^* dx \quad (22)$$

so that

$$P_{m1} = \operatorname{Re} \left\{ mI^2 (K_1 \bar{Z}_m) \right\} \quad (23)$$

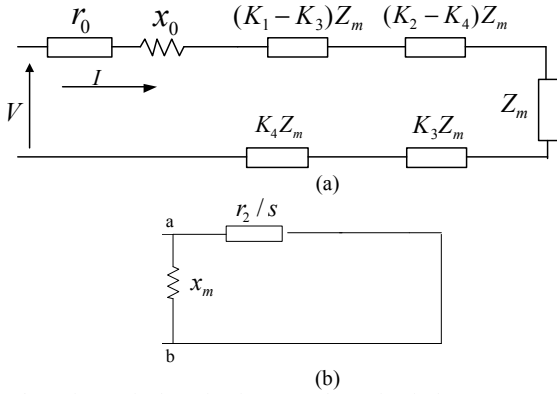


Fig. 4. The series equivalent circuit. (a) Per-phase circuit. (b) Z_m representation

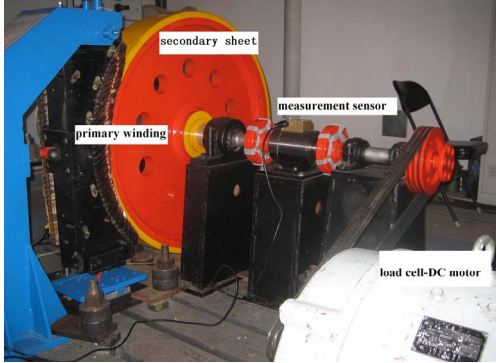


Fig. 5. The SLIM prototype.

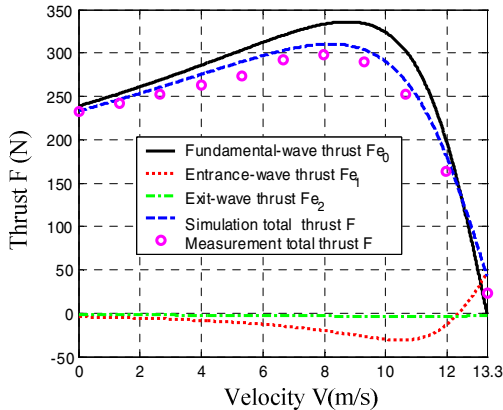


Fig. 6. The thrusts resulted from different air gap flux densities.

where

$$K_1 = -\frac{\alpha_1}{2P} \left(\frac{\tau_e}{\tau \tau_e + j\alpha_1 \pi (\tau - \tau_e)} \right). \quad (24)$$

However, the entrance-wave thrust F_{e1} can be used to derive how much of P_{m1} is converted into mechanical power. This is similar to (20):

$$F_{e1} = \frac{l_w}{2} \operatorname{Re} \int_0^{L_p} \bar{B}_1 e^{-\frac{x}{\alpha_1} - j\frac{\pi x}{\tau_e}} \left\{ \bar{J}_1 e^{-j\frac{\pi x}{\tau}} \right\}^* dx = \frac{mI^2 \operatorname{Re} \{ K_3 \bar{Z}_m \}}{v_s} \quad (25)$$

which becomes

$$P_{e1} = mI^2 \operatorname{Re} \{ (1-s) K_3 \bar{Z}_m \} \quad (26)$$

where

$$K_3 = \frac{-1}{2P} \cdot \frac{1}{\tau \tau_e + j\alpha_1 \pi (\tau - \tau_e)} \cdot \frac{\tau (\tau_e + j\alpha_1 \pi)}{j\pi}. \quad (27)$$

The total input impedance of the entrance wave is $K_1 Z_m$. For (26), the thrust (main) impedance is $K_3 Z_m$. Therefore the impedance (which does not contribute to the thrust) can be defined as $(K_1 - K_3) Z_m$, this is called the entrance-wave leakage impedance. These are illustrated in the series equivalent circuit in Fig. 4(a).

C. Exit-wave Impedance

The exit-wave electromagnetic power P_{m2} for the exit-end flux density wave can be obtained in a similar manner:

$$P_{m2} = mI^2 (K_2 \bar{Z}_m) \quad (28)$$

where the constant is

$$K_2 = -\frac{\alpha_2}{2P} \frac{\tau_e}{\tau \tau_e + j\alpha_2 \pi (\tau + \tau_e)}. \quad (29)$$

The exit-wave thrust F_{e2} is

$$F_{e2} = \frac{mI^2 \operatorname{Re} \{ K_4 \bar{Z}_m \}}{v_s} \quad (30)$$

where

$$K_4 = -j \frac{\alpha_1}{2P} \frac{\tau (\tau_e + j\alpha_2 \pi)}{\alpha_2 \pi [\tau \tau_e + j\alpha_2 \pi (\tau + \tau_e)]}. \quad (31)$$

Again, we can divide the impedances into two. The exit-wave equivalent impedance is $K_2 Z_m$ but the exit-wave main impedance is $K_4 Z_m$, while the exit-wave leakage impedance is $(K_2 - K_4) Z_m$. These are illustrated in Fig. 4(a).

D. Equivalent Circuit and Calculation

The equivalent circuit is shown in Fig. 4. From (24), (27), (29) and (31) it can be shown (after considerable manipulation) that

$$K_1 + K_2 = K_3 + K_4 = \frac{-[(\alpha_1 + \alpha_2) \tau_e^2 + j2\pi \alpha_1 \alpha_2 \tau \tau_e]}{2P [\tau \tau_e + j\alpha_1 \pi (\tau - \tau_e)] [\tau \tau_e + j\alpha_2 \pi (\tau + \tau_e)]} \quad (32)$$

This illustrates that the sum of entrance and exit leakage impedances is zero because the total secondary input power ($K_1 + K_2$ term) is equal to the total mechanical output power ($K_3 + K_4$ term). Therefore the total equivalent longitudinal end-effect impedance Z_L is

$$\bar{Z}_L = K_L \bar{Z}_m \quad (33)$$

where

$$K_L = K_1 + K_2 \quad (34)$$

It is necessary to adjust the z -axis current in the secondary conducting sheet and y -axis flux density in the air-gap. These corrections are called the transverse different-width correction coefficient K_t and the air-gap flux density correction coefficient K_b . These are analogous to end-ring and fringing effects in rotating induction motors. The half-filled slots at the ends of the machine are accounted for by use of the half-filled slot correct coefficient K_p . These coefficients lead to small corrections in the fundamental and longitudinal end-effect impedances which make the performance analysis more accurate [13].

The modified fundamental impedance Z_{mc} is

$$\bar{Z}_{mc} = K_p K_t K_b \bar{Z}_m \quad (35)$$

and modified longitudinal end-effect impedance Z_{Lc} becomes

$$\bar{Z}_{Lc} = K_p K_L \bar{Z}_m. \quad (36)$$

This leads to the total impedance Z_t where

$$\bar{Z}_t = r_0 + jx_0 + \bar{Z}_{mc} + \bar{Z}_{Lc} \quad (37)$$

and the total thrust F is modified so that

$$F = \frac{mI^2 \operatorname{Re}\{\bar{Z}_{mc} + \bar{Z}_{Lc}\}}{v_s}. \quad (38)$$

The power factor $\cos \phi$ is important in induction machines and this can be obtained from

$$\cos \phi = \cos \left[\tan^{-1} \left(\frac{\operatorname{Im}(\bar{Z})}{\operatorname{Re}(\bar{Z})} \right) \right] \quad (39)$$

with the efficiency η calculated from

$$\eta = \operatorname{Re}\{\bar{Z}_{mc} + \bar{Z}_{Lc}\} \frac{(1-s)}{\operatorname{Re}\{\bar{Z}_t\}}. \quad (40)$$

IV. SIMULATION AND EXPERIMENTATION

In order to validate the series equivalent circuit analysis developed in the previous section, the algorithm was simulated and experimental verification obtained from a SLIM prototype. The test rig is shown in Fig. 5. A rotating copper wheel arrangement was used to represent an infinitely long secondary sheet. The dimensions of the machine are given in Table I. The machine was a 4 kW 220 V 3-phase 50 Hz 6-pole linear induction machine with a synchronous speed of 13.3 m/s at 50 Hz. The supply was a variable-frequency inverter source so that a set of torque/speed curves could be obtained for a variety of frequencies.

TABLE I
THE DIMENSIONS OF THE SLIM

Pole pitch [m]	0.113
Number of pole pairs	3
Secondary width [m]	0.17
Secondary sheet thick [mm]	3
Frequency [Hz]	50
Thrust [N]	330
Primary length/width [m]	0.45/0.1
Air gap length [mm]	7
Primary phase voltage [V]	220

The machine was loaded with a DC machine connected to the SLIM via belts. The DC machine could operate at any desired speed including operation at the synchronous speed (driven by the load in motoring mode). The system could measure the SLIM velocity, load power and thrust. Other variables such as

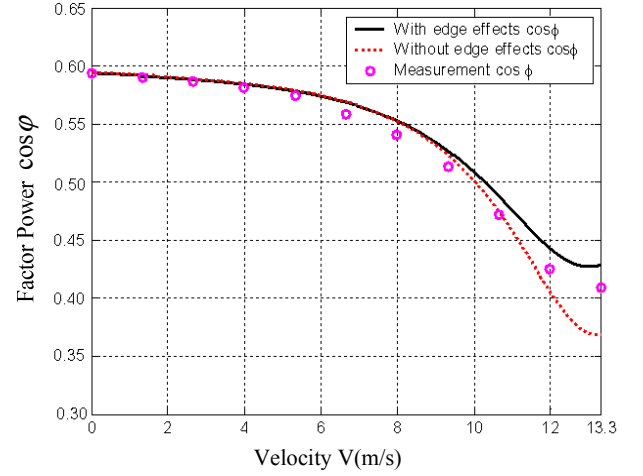


Fig. 7. The power factor curves.

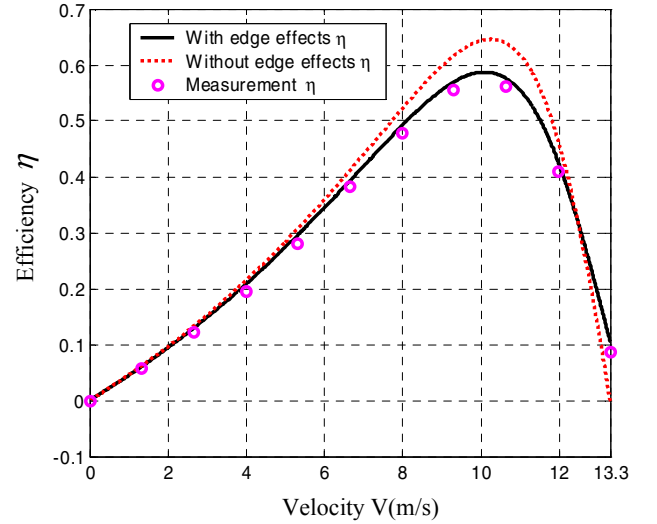


Fig. 8. The efficiency curves.

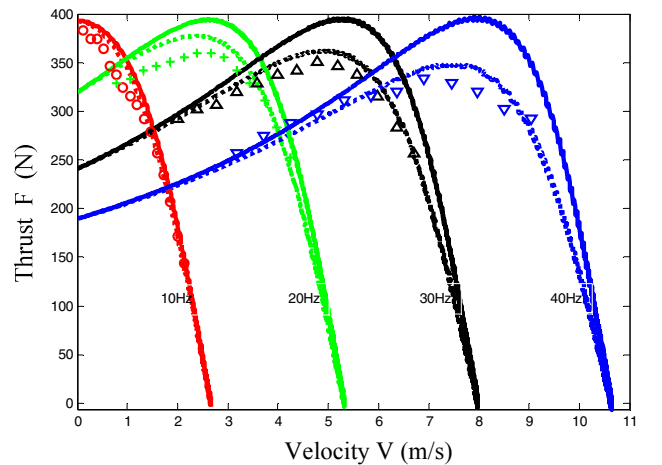


Fig. 9. The thrust curves in constant current and different frequencies.

the primary phase current and phase voltage were recorded using a power analyzer [10].

The simulation and experimental results for the thrust, power factor, and efficiency are given in Figs. 6, 7 and 8 for a constant input primary voltage of 220 V and a frequency of 50 Hz. The thrust/velocity curves shown in Fig. 6 are very similar to those of a rotating cage induction machine. The entry-wave thrust is negative except over a small slip region. The exit-wave thrust is very small over the whole operational region. The total calculated thrust F curve, which is the sum of the three derived thrust components, agrees well with the measured points. The net thrust curve is lower than the fundamental curve due to the entry-wave thrust (apart from close to synchronism where the entry-thrust becomes positive). This illustrates that consideration of the fundamental flux wave only is insufficient for accurate analysis. In Figs. 7 and 8, the power factor and efficiency are influenced little by the end effects in the lower speed region. As the velocity increases and the entry-wave thrust becomes positive near the synchronous speed, the end effects do begin to affect the power factor and efficiency to a greater extent. These simulations are consistent with the results found in [11]. A peak efficiency of about 60 % was found. While this seems low it is reasonable for a small robust induction machine.

Fig. 9 shows the thrust curves with a constant primary current of 15 A at varying frequencies from 0 to 40 Hz. The solid lines represent the thrusts without end effects in the simulation, the dashed lines represent the thrusts with end effects in the simulation and the markers represent the measured. The thrusts at low velocity (high slip) are not affected by the end effects. As the speed increases then end effects have an increasing influence on the thrust across all the different constant-frequency curves.

V. CONCLUSIONS

This paper uses an equation for the air-gap flux density and develops a convenient series equivalent circuit to calculate the performance, including thrust, efficiency and power factor, for a single-sided induction machine. The model is comprehensive and considers the longitudinal end-effects, transverse edge-effects and half-filled slots. Test data from a 4 kW SLIM prototype is put forward to validate the simulation results and good correlation was found. This model could also be implemented to simulate high power SLIMs and would be useful for designing linear machines for use in applications such as a rapid transport or even low-speed low-head hydro water wheels where it would be used in generating mode.

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APPENDIX

List of principal symbols

σ	Conductivity of secondary sheet
ω	Angular frequency of power supply
τ_e	Half-wave length of end-effect wave
α_1	Length of entry-end-effect wave penetration coefficient
α_2	Length of exit-end-effect wave penetration coefficient
e_0	Primary phase induced electromotive force
τ	Pole pitch in m (or half wave length)
τ_e	End-effect half wave length
b_y	y-axis magnetic flux density
g	Equivalent air-gap length
g_m	Mechanical air gap length
l_w	Width of primary lamination
m	Phase number
s	Per-unit slip
v_s	Motor synchronous speed
w_1	Number of turns per phase for primary winding
B_0	Amplitude for magnetic flux density of normal wave
B_1	Amplitude for magnetic flux density of entry-end-effect wave
B_2	Amplitude for magnetic flux density of exit-end-effect wave
E_0	Primary induced electromotive force per phase by normal electric field intensity
E_1	Primary induced electromotive force per phase by entry-end electric field intensity
E_2	Primary induced electromotive force per phase by exit-end electric field intensity
F_{e0}	Fundamental-wave thrust
F_{Ω}	Fundamental-wave mechanical thrust
F_{e1}	Entrance-wave thrust
F_{e2}	Exit-wave thrust
G	Goodness factor
J_1	Equivalent primary current sheet
L_p	Primary length
K_t	Transverse end correct coefficient
K_b	Air gap flux density correct coefficient
K_p	Half-filled slot correct coefficient
K_{w1}	Winding coefficient of primary winding
P_{m1}	Electromagnetic power by entrance flux density wave
P_{m2}	Electromagnetic power by exit-end flux density wave
X_m	Phase magnetizing reactance
Z_m	Primary Phase magnetizing impedance
Z_L	Total equivalent longitudinal end-effect impedance
Z_{Lc}	Correct longitudinal end-effect impedance
Z_{mc}	Correct fundamental impedance
Z_t	Total circuit impedance